

TRT-Remote-OLED

Control Unit



P/N 800ATC-H-(Rxx)-(Rxx)

Operation and Installation

(Dokument-Nr. 03.202.010.71e)

Operation and Installation



Change History

Revision	Date	Description of Change	
1.00	22.10.2012	First Release	
2.00		Change of company name to f.u.n.k.e. AVIONICS GmbH Added information for TSO certification	

List of Service-Bulletins (SB)

Service Bulletins have to be inserted into this manual and to be enlisted in the following table.

SB No	Rev. No.	Issue Date	Insertion Date	Name

Survey of Variants

Part Number	Description
P/N 800ATC-H-(R00)-(R00)	First version with OLED-Display
P/N 800ATC-H-(R00)-(R01)	OLED-Version with TSO-Certification





Table of Contents

1	GENI	ERAL	4
	1.1	Symbols	4
	1.2	Abbreviations	5
	1.3	Customer Support	6
	1.4	Features	6
	1.5	Deviations and Important Notes	7
2	OPE	RATION	8
	2.1	Controls	8
	2.2	Switch ON/OFF	10
	2.3	Display - Indications	11
	2.4	Display-Brightness	12
	2.5	Flight-ID (FID)	12
		2.5.1 Display Flight-ID	13
		2.5.2 Configure Flight-ID	13
	2.6	Transponder Mode selection	14
	2.7	Squawk-Setting	14
	2.8	VFR – Squawk	15
	2.9	ID – Special Position Identification (SPI): "Squawk Ident"	15
3	INST	ALLATION	16
	3.1	Notes	16
	3.2	Scope of Delivery	16
	3.3	Unpacking and Inspecting of the Equipment	16
	3.4	Mounting	17
	3.5	Equipment Connections	18
		3.5.1 Electrical Connections	18
	3.6	Wiring	18
		3.6.1 Conductor Cross Section	18
		3.6.2 Pin Assignment	18
	3.7	Post-Installation Check	19
	3.8	Drawings	20
		3.8.1 Dimensions	20
		3.8.2 Mounting Advices	20
4	APPE	ENDIX	21
	4.1	Technical Data	21
	12	Environmental Conditions	22



1 GENERAL

This manual contains information about the physical, mechanical and electrical characteristics and about installation and operation of the transponder remote control TRT800RT.

The conditions and tests required for TSO approval of this article are minimum performance standards. Those installing this article either on or within a specific type or class of aircraft must determine that the aircraft installation conditions are within the TSO standards which include any accepted integrated non-TSO function standards. TSO articles and any accepted integrated non-TSO function(s) must have separate approval for installation in an aircraft. The article may be installed only according to 14 CFR part 43 or the applicable airworthiness requirements. This is an incomplete system (antenna is required) intended to provide the TSO functions specified in this Operation and Installation Manual, section 5.1.

1.1 Symbols



Advices whose non-observance can cause radiation damage to the human body or ignition of combustible materials



Advices whose non-observance can cause damage to the device or other parts of the equipment.



Supplementary information



1.2 Abbreviations

Abb.	Meaning	Explanation
FID	Flight ID	Flight plan number or if not assigned registration number of aircraft
SPI	Special Position Identification	Activation on request by controllers "Squawk Ident", transmits SPI Pulse for 18 seconds, which highlights the respective traffic item on the controllers radar screen



1.3 Customer Support

In order to facilitate a rapid handling of return shipments, please follow the instructions of the input guide "Reshipment RMA" provided at the **Service**-Area within the f.u.n.k.e. AVIONICS GmbH web portal www.funkeavionics.de.



Any suggestions for improvement of our manuals are welcome. Contact: service@funkeavionics.de.



Informations on software updates are available at f.u.n.k.e. AVIONICS GmbH.

1.4 Features

- Remote control and indication of all transponder settings/data during standard operation (e.g. squawk code, operation mode, altitude etc).
- Ideal for tandem seated aircraft
- Use of transponder still possible without activation of remote control
- Automatic deactivation of remote control on deactivation of transponder



1.5 Deviations and Important Notes

The following deviations have been granted by the FAA:



The device is certified to FAA TSO-C112c in lieu of FAA TSO-C112d.



Environmental Qualification Testing was performed according to RTCA/DO-160D in lieu of RTCA/DO-160F.



The Software of this device was developed according to RTCA DO-178B Design Assurance Level "D".



The ADS-B function of this device has not been evaluated during the FAA TSO certification process other than to ensure non-interference. Due to the SW assurance of DAL "D", the ADS-B function cannot be used in the US National Airspace.

Connecting a GPS source will not comply to FAA certification requirements with regard to ADS-B.



This device contains an altitude measuring device that has not been evaluated during the FAA TSO certification process other than to ensure non-interference.

This feature must be evaluated for intended function during FAA installation approval.



Compliance to RTCA DO-181D was verified by testing against EUROCAE ED-73C. The two standards are technically identical with regard to the TRT800H.

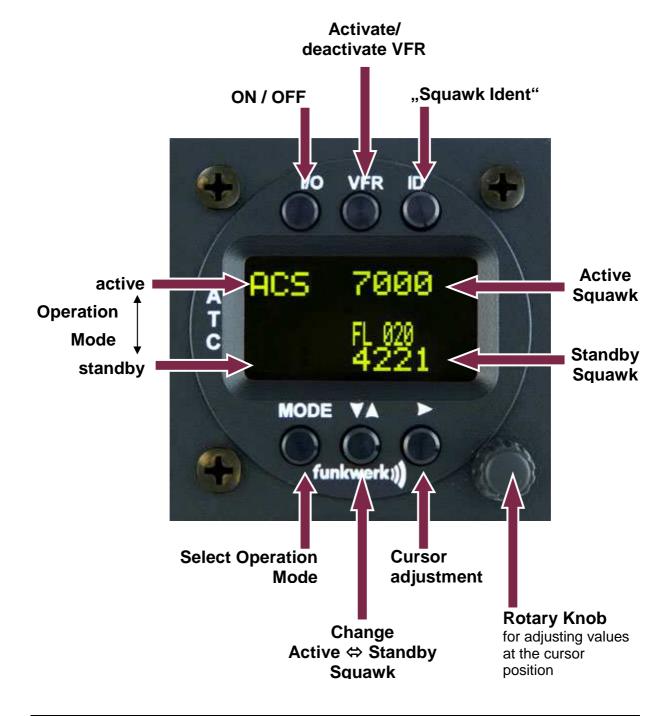


2 OPERATION

2.1 Controls



If remote unit settings are inconsistent with transponder settings as a result of short time transmission problem (e.g. simultaneous user inputs) remote unit settings will be reverted to current transponder setting.





	T	
I/O	ON/OFF	► Switch ON press button for approx. 0,5 s► Switch OFF press button for approx. 3 s
● VFR	VFR	 activate/deactivate VFR Squawk (press shortly) store active Squawk as VFR/VFRW-Squawk (press button 3 s, →2.8)
● ▲ ▼	CHANGE	 change between active and standby-Squawk works as cursor back button (opposite function of the cursor button) during entering values and also for navigating backwards through the configuration menu
● ID	IDENT	 ▶ "Squawk Ident", sends Ident marking (SPI) for 18 s (→ 2.9) ▶ Enter Flight-ID (FID) setup (in standby mode, press button for approx. 5s)
● MODE	MODE	► Select transponder operational mode (→ 2.6)
• ►	CURSOR	► Set position of Cursor
	Rotary Knob	► Adjust/Enter values at current cursor position, select options; set standby Squawk (→ 2.7)

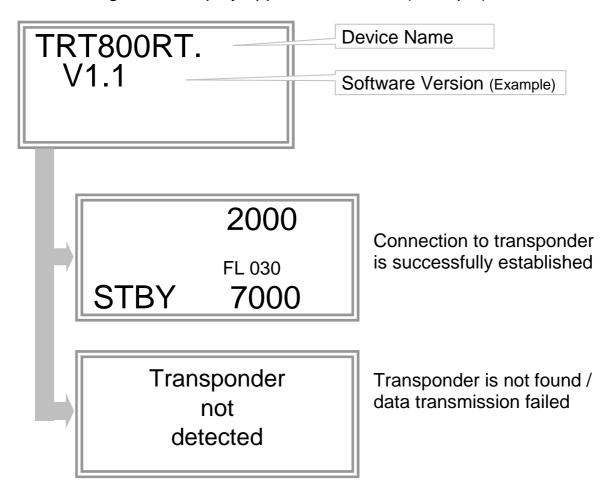


2.2 Switch ON/OFF

Switch **ON**: press **I/O** for 0.5 s Switch **OFF**: press **I/O** for 3 s

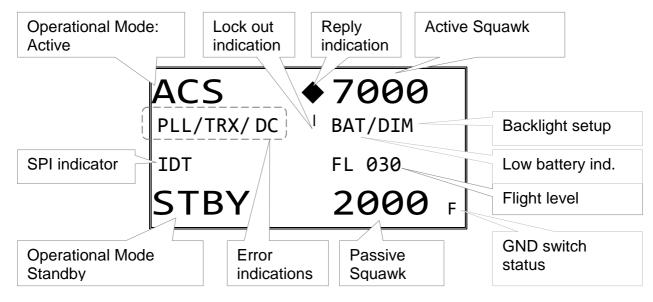
2.2.1 Display information after power up

After turning-on the display appears as follows (example):





2.3 Display - Indications



Value	Meaning	Remarks
7000	active Squawk	
2000	Standby Squawk	Could be changed with active Squawk by pressing ● ▲ ▼
FL 010	Flight Level	Flight Level (in 100 ft steps)
FLerr	Invalid altitude measuring	
ACS	Operational Mode (STBY, A-S, ACS, AC, A)	Modes see section 2.6
IDT	transmits Ident-Marking (SPI)	ID ("Squawk Ident") has been pressed – active for 18 s
•	Transponder replies	No indication on ADSB and
	on Mode-S, Mode-A or Mode-C interrogations	squitter transmission
I	Transponder is locked by a ground station and will be directly addressed	Lock Information (indicated below the diamond)
F	in-flight	Ground-Switch-Info
G	on-ground	(if available)





Error indicators			
PLL	PLL Error	Internal Error	
TRX	Transmit Failure	Check antenna and wiring	
DC	Low internal voltage	Internal error	
FPG	FPGA-Failure	Internal error	
BAT	Battery Power too low	maybe battery/generator fault	

Note: Short time indication symbols (e.g. lock-out indication or reply indication) are displayed/cleared in different time interval or delayed in comparison with TRT800H screen.

2.4 Display-Brightness

In active mode (not standby) press ● ► for 2 s

Display indicates "DIM" → Adjust brightness (DIM) with rotary knob

Return to normal operation: press ● ► or wait 5 s.

2.5 Flight-ID (FID)

The FID is an identifier required by Mode-S Operation. During future application of flight plans a FID could be assigned on a per flight basis. If no FID is assigned (today's normal case) the registration marking of the aircraft should be used as FID. The FID should <u>not</u> contain dashes or blanks.



2.5.1 Display Flight-ID

Press MODE (repeatedly) until "STBY" appears

Press and hold **ID** while a counter is shown beside the active squawk. During a few seconds Flight Identification is displayed



2.5.2 Configure Flight-ID

Press • MODE (repeatedly) until "STBY" appears

Press and hold **ID** while a counter is shown beside the active squawk. Release **ID** when "CHANGE FID" is displayed



Enter Flight-Id with cursor button ● ► and rotary knob



Enter FID <u>left-aligned</u>, <u>without any blanks or dashes (!)</u>, e.g. 12345621DEFAV for the marking D-EFAV. The last remaining digits shall be filled with blanks

Press • MODE to save and return to STBY



After FID is changed: Check that new FID is accepted and correctly set



2.6 Transponder Mode selection

Press **MODE** (repeatedly) to select from following Modes:

- STBY → "Standby"
 Transponder does not respond to any interrogation. Squitter and ADS-B output is not active.
- ACS → "Mode A+C+S"
 Standard condition; transponder responds to mode A, C and S interrogations.
- A-S → "Mode A+S, no C"
 Altitude is not transmitted (neither on C nor on S requests). All other Mode-S data as well as Mode-A replies are transmitted.

If no 24-bit address (AA) was defined or entered as "000000" the transponder works as Mode A-C transponder, in that case the following Modes are possible apart from Standby:

- AC- → "Mode A+C"
 Transponder replies only on Mode A and Mode-C interrogations.
- A-- → "Mode A"
 Transponder replies only on Mode A interrogations.



In STBY (Standby) mode, all transponder transmissions are disabled completely! Therefore, the transponder is not visible in this mode to air traffic control or the anti-collision systems onboard other aircrafts.

Never use the STBY mode in flight unless you are requested to do so by air traffic control. Always remember to put the transponder in active mode prior to take off!

2.7 Squawk-Setting

The active Squawk is displayed in the upper line, while the standby Squawk is presented at the lower line.

Setting the Standby Squawk:

- Press ► to set the cursor ("^"), turn rotary knob to set numbers of the standby Squawk.
- Press ▼ ▲ to activate the Standby Squawk (this moves the current active Squawk into Standby)

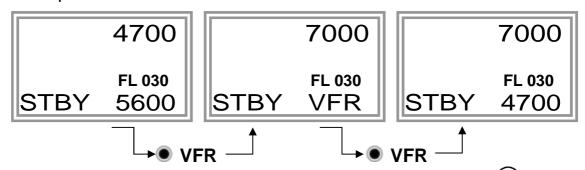


2.8 VFR – Squawk

The transponder features a user-defined squawk code for VFR-flight (factory setting: 7000):

- Activate VFR-Squawk:
 - Press **VFR** ("VFR" is indicated), now the active Squawk is moved into Standby but not visible because the indication of the Standby Squawk is overlapped by "VFR"
- Display Standby Squawk:
 Press ●VFR or ●▼▲ or use the Frotary knob (the VFR-Squawk remains active!)

Example:



- Now the Standby Squawk can be adjusted by using the rotary knob and activated with
 ▼ ▲.
- In order to store the current active Squawk as new VFR-Squawk (replacing the factory setting 7000):
 Press and hold VFR until an "S" is indicated (approx. 3 s); after releasing the button "VFR" is shown

2.9 ID – Special Position Identification (SPI): "Squawk Ident"

Press ID to activate transmission of the special position identification pulse with every reply within 18 seconds; "IDT" appears on the display

By pressing **ID** a special position identification pulse (SPI) is transmitted with every reply within 18 seconds, which causes an accented marking on the Controller's screen. The "Special Position Identification" has to be activated after the "Squawk Ident" request of the Controller.



3 INSTALLATION

3.1 Notes

The following suggestions should be considered before installing.

The assigned installation company will supply wiring. For diagrams refer to 3.6 Wiring.

Remote Header and cables shall be installed according to "FAA AC-143.13-2A Acceptable Methods, Techniques and Practices – Aircraft Alterations" and the appropriate manufacturer's instructions.

3.2 Scope of Delivery

Item name / number	Description
TRT800RT	Remote Header TRT800H-Remote Control Unit
ATR600RTK3	Connector
M4X8ZSW (3 pieces)	screw for panels up to 5 mm thickness
03.202.010.71e	Manual "Operation and Installation"
	EASA Form 1

3.3 Unpacking and Inspecting of the Equipment

Carefully unpack the equipment and inspect for transport damages. If a damage claim has to be filed, save the shipping container and all packing materials as evidence to your claim.



For storage or reshipment the original packaging should be used.



3.4 Mounting

- In cooperation with a maintenance shop, location and kind of the installation are specified. The maintenance shop can supply all cables. Suitable sets of cables are available from f.u.n.k.e. AVIONICS GmbH.
- Select a position away from heat sources. Care for adequate convection cooling.
- Leave sufficient space for the installation of cables and connectors.
- Avoid sharp bends and wiring close to control cables.
- Leave sufficient lead length for inspection or repair of the wiring of the connector (containing the memory), so that when the mounting hardware for the rear connectors is removed, the assembly may be pulled forward several inches.
- Bend the harness at the rear connectors to inhibit water droplets (formed due to condensation) from collecting in the connector.
- For mounting details/drawing refer to chapter 3.8.2 Mounting Advices.



3.5 Equipment Connections

3.5.1 Electrical Connections

One 9 pin D-SUB miniature connector includes all electrical connections.



When powering the remote control not by the output of the switched power output of the remote controlled device but directly by the aircraft's power supply, the power input line (+UB) must be equipped with an external fuse (1Amp./slow).

3.6 Wiring

3.6.1 Conductor Cross Section

Power Supply (Power, GND): AWG20 (0,62 mm²) Signals: AWG22 (0,38 mm²)

The conductors must be approved for aircraft use. RX and TX should be shielded individually, or at least be pair-twisted and shielded together.

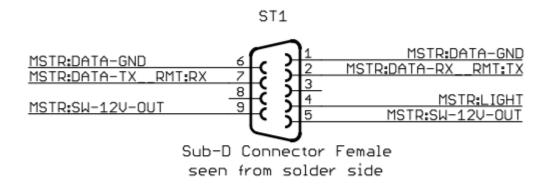
3.6.2 Pin Assignment

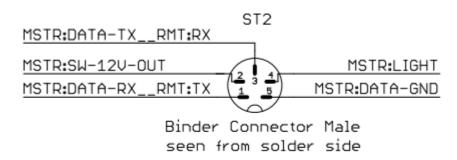
Pin	Signal	Remarks
1	GND (Data)	-
2	Data TX	Output
3	-	
4	Light	Input
5	UB+	Power input
6	GND (Data)	
7	Data RX	Input
8	-	-
9	UB+	Power input ⁽¹⁾



(1) UB+: Can be connected to switched power output of transponder.







3.7 Post-Installation Check



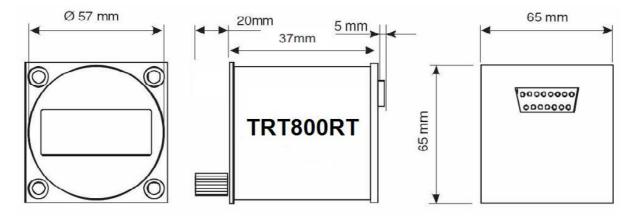
A certified maintenance shop must verify proper operation of remote control header".

All steering and control functions of the aircraft are to be examined, in order to exclude disturbances by the wiring.

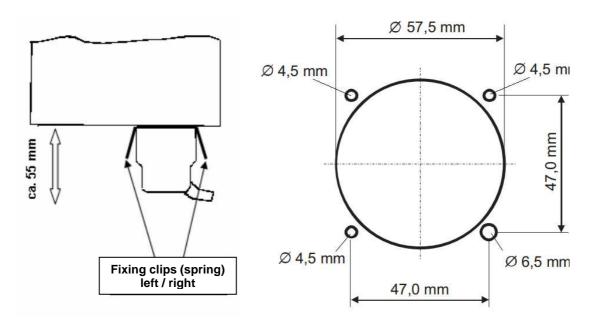


3.8 Drawings

3.8.1 Dimensions



3.8.2 Mounting Advices





Connector (plug) has to be clamped with both spring locks. New screws may be turned in more than max. 15mm into the device – even if no hard limit is noticeable.



4 APPENDIX

4.1 Technical Data

Compliance	CS ETSO 201120
Compliance	CS-ETSO-2C112a
	EASA.21O.269
	FAA TSO-C112c
Applicable Documents	CS-ETSO-2C112a
	FAA TSO-C112c
	EUROCAE ED-73C Class 1 Level 2es
	RTCA DO-181D Class 1 Level 2es
	EUROCAE ED-26
	RTCA DO-160D
	RTCA DO-178B Software-Level D
Temperature Ranges	
Operation	-20 °C to +55 °C; for 30 min +70°C
Storage	-55 °C to +85 °C
Altitude Range	≤ 35 000 ft
Shock	6 G Operation
	20 G crash safety
Environmental Categories	RTCA DO-160D Env.Cat.: [C1Z]CAA[SM]XXXXXXZBAAA[TT]M[B3F3]XXA
Power Supply	13,8 VDC (10 VDC 16 VDC)
	0,06 A @ 13,8 VDC (typ.)
	600 mW (max)
Fuse	External 1A-slow-blow fuse
Mounting	Panel cut-out d=57,5 mm
Weight	0,2 kg (0.43 lbs.)



4.2 Environmental Conditions

Characteristic DO-160D	Section	Cat.	Condition	
Temperature / Altitude	4.0			
Low ground survival temperature	4.5.1	-	– 55°C	
Low operating temperature	4.5.1		– 20°C	
High ground survival Temperature	4.5.2	C1	+ 85°C	
High Short-time Operating Temperature	4.5.2		+ 70°C	
High Operating Temperature	4.5.3		+ 55°C	
In-Flight Loss of Cooling	4.5.4	Z	No auxiliary cooling required	
Altitude	4.6.1	C1	35 000 ft	
Temperature Variation	5.0	С	2°C change rate minimum per minute	
Humidity	6.0	Α		
Shock	7.0	A	6 G operational shocks 20 G Crash Safety Test Type R in all 6 directions	
Vibration	8.0	S	Vibration Curve M	
Explosion Proofness	9.0	Χ	No test required	
Water Proofness	10.0	Χ	No test required	
Fluids Susceptibilities	11.0	Χ	No test required	
Sand and Dust	12.0	Χ	No test required	
Fungus Resistance	13.0	X	No test required	
Salt Spray	14.0	Χ	No test required	
Magnetic Effect	15.0	Z	Less than 0,3 m	
Power Input (DC)	16.0	В		
Voltage Spike Conducted	17.0	А		
Audio Frequency Conducted Susceptibility	18.0	А		





Characteristic DO-160D	Section	Cat.	Condition
Induced Signal Susceptibility	19.0	Α	
Radio Frequency Susceptibility	20.0	TT	
Emission of RF Energy	21.0	М	
Lightning Induced Transient Susceptibility	22.0	B3F3	
Lightning Direct Effects	23.0	Χ	No test required
Icing	24.0	Χ	No test required
Electrostatic Discharge (ESD)	25.0	Α	

Operation and Installation



N	otes	•
ıv	$O(C_0)$	_







f.u.n.k.e. AVIONICS GmbH

Heinz-Strachowitz-Str. 4 DE-86807 Buchloe Germany

phone.: +49-8241 80066 0 fax.: +49-8241 80066 99

E-mail:

service@funkeavionics.de www.funkeavionics.de